Designation: D6742/D6742M – 23

Standard Practice for Filled-Hole Tension and Compression Testing of Polymer Matrix Composite Laminates[1]

This standard is issued under the fixed designation D6742/D6742M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

1. Scope

1.1 This practice provides instructions for modifying open-hole tension and compression test methods to determine filled-hole tensile and compressive strengths. The composite material forms are limited to continuous-fiber reinforced polymer matrix composites in which the laminate is both symmetric and balanced with respect to the test direction. The range of acceptable test laminates and thicknesses are described in 8.2.1.

1.2 This practice supplements Test Methods D5766/D5766M (for tension testing) and D6484/D6484M (for compression testing) with provisions for testing specimens that contain a close-tolerance fastener or pin installed in the hole. Several important test specimen parameters (for example, fastener selection, fastener installation method, and fastener hole tolerance) are not mandated by this practice; however, repeatable results require that these parameters be specified and reported.

1.3 The values stated in either SI units or inch-pound units are to be regarded separately as standard. The values stated in each system may not be exact equivalents; therefore, each system shall be used independently of the other. Combining values from the two systems may result in non-conformance with the standard.

1.3.1 Within the text the inch-pound units are shown in brackets.

1.4 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety, health, and environmental practices and determine the applicability of regulatory limitations prior to use.*

1.5 *This international standard was developed in accordance with internationally recognized principles on standardization established in the Decision on Principles for the Development of International Standards, Guides and Recommendations issued by the World Trade Organization Technical Barriers to Trade (TBT) Committee.*

2. Referenced Documents

2.1 *ASTM Standards:*[2]

D792 Test Methods for Density and Specific Gravity (Relative Density) of Plastics by Displacement

D883 Terminology Relating to Plastics

D3171 Test Methods for Constituent Content of Composite Materials

D3878 Terminology for Composite Materials

D5229/D5229M Test Method for Moisture Absorption Properties and Equilibrium Conditioning of Polymer Matrix Composite Materials

D5766/D5766M Test Method for Open-Hole Tensile Strength of Polymer Matrix Composite Laminates

D6484/D6484M Test Method for Open-Hole Compressive Strength of Polymer Matrix Composite Laminates

D6507 Practice for Fiber Reinforcement Orientation Codes for Composite Materials

D8509 Guide for Test Method Selection and Test Specimen Design for Bolted Joint Related Properties

E6 Terminology Relating to Methods of Mechanical Testing

E177 Practice for Use of the Terms Precision and Bias in ASTM Test Methods

E456 Terminology Relating to Quality and Statistics

3. Terminology

3.1 *Definitions*—Terminology D3878 defines terms relating to high-modulus fibers and their composites. Terminology D883 defines terms relating to plastics. Terminology E6 defines terms relating to mechanical testing. Terminology E456 and Practice E177 define terms relating to statistics. In the event of a conflict between terms, Terminology D3878 shall have precedence over the other standards.

3.2 *Definitions of Terms Specific to This Standard*—Refer to Guide D8509.

[1] This practice is under the jurisdiction of ASTM Committee D30 on Composite Materials, and is the direct responsibility of Subcommittee D30.05 on Structural Test Methods.

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[2] For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

3.3 *Symbols:*

A = cross-sectional area of a specimen

d = fastener diameter

D = specimen hole diameter

d_{csk} = countersink depth

d_{fl} = countersink flushness

f = distance, perpendicular to loading axis, from hole edge to closest side of specimen

F_x^{fhcu} = ultimate filled-hole compressive strength in the test direction

F_x^{fhtu} = ultimate filled-hole tensile strength in the test direction

g_1 = distance, parallel to loading axis, from hole edge to end of specimen

h = specimen thickness

P^{max} = maximum force carried by test specimen prior to failure

w = specimen width

4. Summary of Practice

4.1 *Filled-Hole Tensile Strength*—In accordance with Test Method D5766/D5766M, but with a close-tolerance fastener or pin installed in the hole, perform a uniaxial tension test of a balanced, symmetric laminate with a centrally located hole.

4.2 *Filled-Hole Compressive Strength*—In accordance with Test Method D6484/D6484M, but with a close-tolerance fastener or pin installed in the hole, perform a uniaxial compression test of a balanced, symmetric laminate with a centrally located hole.

NOTE 1—For both test methods, ultimate strength is calculated based on the gross cross-sectional area, disregarding the presence of the filled hole. While the filled hole causes a stress concentration and reduced net section, it is common aerospace practice to develop notched design allowable strengths based on gross section stress to account for various stress concentrations (fastener holes, free edges, flaws, damage, and so forth) not explicitly modeled in the stress analysis.

5. Significance and Use

5.1 Refer to Guide D8509.

6. Interferences

6.1 Refer to Guide D8509.

7. Apparatus

7.1 *General Apparatus*—General apparatus shall be in accordance with Test Methods D5766/D5766M (for tension tests) and D6484/D6484M (for compression tests), although with a fastener or pin installed in the specimen hole. The micrometer or gauge used shall be capable of determining the hole and fastener diameters to ±8 µm [±0.0003 in.].

7.2 *Fastener*—The fastener or pin type shall be specified as an initial test parameter and reported. The nominal fastener diameter shall be 6 mm [0.25 in.], unless a range of diameters is being investigated. Some fastener types (for example blind bolts) may not be available in this diameter; for these, it is recommended to use a fastener for which the diameter is as close as possible to 6 mm [0.25 in.]. The installation torque (if applicable) shall be specified as an initial test parameter and reported. This value may be a measured torque or a specification torque for fasteners with lock-setting features. If washers are used, the washer type, number of washers, and washer location(s) shall be specified as initial test parameters and reported. Reuse of fasteners is not recommended because of potential differences in through-thickness clamp-up for a given torque level, caused by wear of the threads.

7.3 *Torque Wrench*—If using a torqued fastener, the torque wrench used to tighten the fastener shall be capable of determining the applied torque to within ±10 % of the desired value.

8. Sampling and Test Specimens

8.1 *Sampling*—For tension tests, sampling shall be in accordance with Test Method D5766/D5766M. For compression tests, sampling shall be in accordance with Test Method D6484/D6484M.

8.2 *Geometry:*

8.2.1 *Stacking Sequence*—The standard laminates shall have multidirectional fiber orientations (fibers shall be oriented in a minimum of two directions) and balanced and symmetric stacking sequences. For tension specimens, nominal thickness shall be 2.5 mm [0.10 in.], with a permissible range of 2 mm to 4 mm [0.080 in. to 0.160 in.], inclusive. For compression specimens, nominal thickness shall be 4 mm [0.160 in.], with a permissible range of 3 mm to 5 mm [0.125 in. to 0.200 in.], inclusive. Fabric laminates containing satin-type weaves shall have symmetric warp surfaces, unless otherwise specified and noted in the report.

NOTE 2—Typically, a $[45_i/-45_i/0_j/90_k]_{ns}$ tape or $[45_i/0_j]_{ns}$ fabric laminate should be selected such that a minimum of 5 % of the fibers lay in each of the four principal orientations. This laminate design has been found to yield the highest likelihood of acceptable failure modes. Consult Practice D6507 for information on fiber orientation codes.

8.2.2 *Specimen Configuration*—For tension tests, the test specimen configuration shall be in accordance with Test Method D5766/D5766M. For compression tests, the test specimen configuration shall be in accordance with Test Method D6484/D6484M. The nominal hole diameter may vary from that specified in Test Methods D5766/D5766M and D6484/D6484M depending upon the type of fastener used.

8.3 *Specimen Preparation*—For tension tests, specimens shall be prepared in accordance with Test Method D5766/D5766M. For compression tests, specimens shall be prepared in accordance with Test Method D6484/D6484M. Use appropriate hole preparation procedures specified by the test requestor.

8.4 If specific gravity, density, reinforcement volume, or void volume are to be reported, then obtain these samples from the same panels being tested. Specific gravity and density may be evaluated by means of Test Method D792. Volume percent of the constituents may be evaluated by one of the matrix digestion procedures of Test Method D3171, or, for certain reinforcement materials such as glass and ceramics, by the matrix burn-off technique of Test Method D3171.